



Advancements in Human-Carrying Quad Copter

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ABSTRACT :

This study presents the design and development of a human-carrying quadcopter, an innovative aerial vehicle aimed at addressing the growing demand for urban air mobility solutions. As cities become increasingly congested, the need for efficient transportation options is more critical than ever. The proposed quadcopter integrates advanced engineering principles, cutting-edge materials, and robust safety features to ensure reliable and safe human transport. This research outlines the aerodynamic design, structural analysis, and propulsion systems utilized to achieve optimal lift and stability while accommodating passenger weight. Key challenges, such as flight control, stability management, and regulatory compliance, are also addressed, highlighting the importance of advanced control algorithms and real-time monitoring systems. Preliminary flight tests demonstrate the quadcopter's performance capabilities, including takeoff, landing, and maneuverability, establishing a foundation for future development. The findings of this study not only contribute to the evolving field of aerial transportation but also offer insights for further innovations in human-carrying UAV technology, paving the way for practical applications in urban environments and emergency response scenarios.

Keywords: *Quadrotorhelicopter or Quadrotor, drone's, vertically oriented propellers*

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INTRODUCTION

The advent of unmanned aerial vehicles (UAVs) has revolutionized various sectors, including logistics, surveillance, and agriculture. Among these innovations, human-carrying quadcopters represent a groundbreaking development in urban air mobility, providing a potential solution to the increasing congestion and inefficiencies of traditional ground transportation systems. As urban populations grow, so does the demand for rapid and efficient transit options. This has sparked interest in aerial transportation technologies that can alleviate traffic woes and reduce travel times, while also offering an eco-friendly alternative to conventional vehicles.

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The human-carrying quadcopter is designed to transport passengers in a safe and efficient manner, leveraging the versatility of quadcopter design principles to achieve vertical takeoff and landing (VTOL) capabilities. This aircraft combines advanced aerodynamics, lightweight materials, and state-of-the-art propulsion systems to ensure optimal performance and stability. Moreover, incorporating autonomous flight capabilities and intelligent navigation systems enhances its operational safety and efficiency, addressing some of the primary concerns associated with manned aerial vehicles.

Despite the promising potential of human-carrying quadcopters, several challenges

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remain to be addressed. These include ensuring compliance with aviation regulations, developing reliable safety systems, and overcoming technical barriers related to flight control and battery life. This study aims to explore these challenges while presenting a comprehensive design and development framework for a human-carrying quadcopter. By investigating aerodynamic performance, structural integrity, and propulsion efficiency, this research seeks to provide valuable insights into the practical implementation of this innovative aerial transportation solution.

Through a systematic approach that includes theoretical modeling, simulation, and preliminary flight testing, this study endeavors to establish a foundation for the future of urban air mobility. The findings will not only advance the understanding of human-carrying quadcopter technology but also contribute to the broader discourse on sustainable transportation solutions, positioning aerial vehicles as a viable option in the evolution of urban transit systems.

1.1 Scope of Project :

The purpose of this project is to engineering knowledge to develop a UAV at low cost based on specifications as finalized in SOR. project will engage the members through proper design cycle which will include the implementation of the aircraft design knowledge, as well as other courses, use of the various CAD software that they have learnt for design and analysis of their project and make critical decisions regarding their design observing its performance as well as their constraints such as budget, manufacturability Etc.

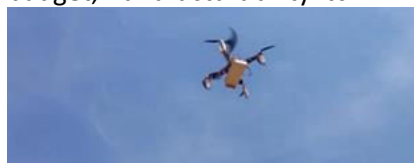


Figure1: Quadcopter during transition

LITERATURE SURVEY

In this section the literature survey of weight carrying quadcopters is discussed. The experimentations conducted by various researchers by influencing the unmanned aerial vehicles (

UAV) process parameters on specific weight carrying method's.

2.1 Quadcopter history:

Etienne Oehmichen was the first scientist who experimented with rotorcraft designs in the 1920s. Among these six designs he tried, his second multicopter had four rotors and eight propellers, all driven by a single engine. The Oehmichen used a steel-tube frame, with two-bladed rotors at the ends of the four arms. The angle of these blades could be varied by warping. Five of the propellers, spinning in the horizontal plane, stabilized the machine laterally. Another propeller was mounted at the nose for steering. The remaining pair of propellers was for forward propulsion.

METHODOLOGY

In a quadcopter drone, two of the motors rotate in a clockwise direction and the other two motors in an anticlockwise direction. The speed of the motors is controlled by the electronic speed controller. If the two motors on the rear side of the drone rotate at high speeds, then the drone moves in a forward direction. If the two motors on the front side of the drone rotate at high speeds, then the drone moves in a backward direction. If the two motors on the left side of the drone rotate at high speeds, then the drone moves in a rightward direction.

Introduction with System Block Diagram:

A block diagram is a high-level and specialized flowchart used in engineering. It is used to describe and improve existing systems or to design new ones. The block diagram's structure provides a high-level overview of important working relationships, key process participants, and major system components. Below is the overview of the major quadcopter components represented as a block diagram

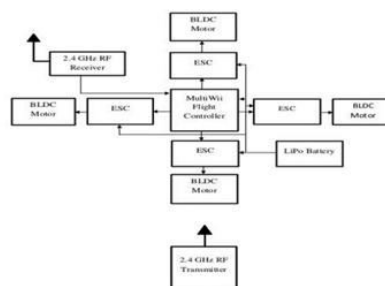


Figure2: Block diagram of a Quadcopter

Technical Specification:



The drone's technical specification is nothing more than a document that through technical data makes a clarified description for the drone's specific use, functionality, or performance



Figure 3: Block diagram of a drone's specific use, functionality

Designing of Quadcopter body:

A quadcopter is a multi-rotor drone that has four arms having a brushless motor on each arm. Quadcopter drone arms are designed in fusion 360 software. Hence this is a delivery drone, the middle part of the body has to be strong to carry the loads. So, a sheet metal body has to be used for the upper and bottom parts of the drone connecting four arms. The drone planned for the current work is initially modeled in fusion 360 software. Figure 2 shows the design frame of the drone in fusion 360 software.



Figure 4 Modeling and rendering of a quadcopter drone in fusion 360

Fabrication:

3D printing is also known as additive manufacturing, and it is the process of building a 3D object from a CAD model. The arms of the quadcopter drone are designed in fusion 360 software. Additive manufacturing of quadcopter arms by using PLA (Polylactic Acid) filament in 3D printing. Repetier Host software is used for slicing and Creality Ender 3.0 is used for printing parts.

Parameter	Value
Shell Thickness	2 mm
Top and Bottom Thickness	2 mm
Infill Pattern	Grid
Support Pattern	Grid
Fill Amount	15%
Print Speed	25 mm/s
Travel Speed	60 mm/s
Layer Thickness	0.2 mm

Table: 1 Printing parameters
Assembly and Flight Testing:

3D printed arms of the quadcopter are assembled to the sheet metal body of the drone. Figure shows the fabrication of the sheet metal body. A brushless DC motor is connected to the electronic speed controller (ESC). Motors are placed on the edge of the arms and ESC is placed in the middle of the arms. A power distribution board is used to distribute the power to all components. APM flight controller is placed in the middle of the drone and it's connected to the receiver, ESC, GPS module, and power module.



Figure 5: Final assessment of the robot

Hardware components: Most of the parts on a quadcopter are dependent on each other and when choosing one part it puts a new set of demands on another. The one most important characteristic to consider is the weight of the quadcopter, since that in turn puts demand on all parts except the ones handling calculations for navigation.

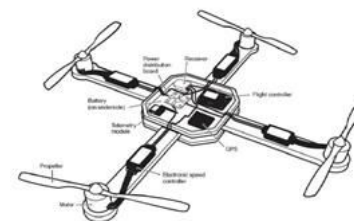


Figure 6: Overall structure of the drone

1. **Motor:** The motors together with the propellers are needed to provide enough thrust to lift the quadcopter with the gripping module and a payload of up to 20% of the total weight. Since the aim is to follow a 2:1 thrust to weight ratio, the total minimum thrust needed to beat at least 1.5 kg.

PROPELLORS:

A propeller is a type of fan that transmits power by converting rotational motion into thrust. A pressure difference is produced between the forward and rear surfaces of the airfoil-shaped blade, and a fluid (such as air or water) is accelerated behind the blade. Propeller dynamics can be modeled by



both Bernoulli's principle and Newton's third law.

It is also the main part of the quad copter for flying, there are two types of propellers used in the quadcopter they mostly left-hand propellers and right-hand propellers. 25 Left hand propellers are also called normal propellers and they are mounted to the motor which is moving in counter-clockwise direction.

MULTI-ROTOR CONTROL BOARD

The KK.2 multi-controller is a flight control board for remote control multi-copters with 2, 3, 4, and 6 rotors. Its purpose is to stabilize the aircraft during flight. To do this it takes the signal from the three gyros on the board (roll, pitch and yaw) and feeds the information into the Integrated Circuit (Atmega IC). This then processes the information according to the KK software and sends out a control signal to the Electronic Speed Controllers (ESCs) which are plugged onto the board and also connected to the motors.

Specifications:-

Size: 50.5mm x 50.5mm x 12mm Weight: 21 gram (Inc Piezo buzzer) IC: Atmega324 PA Gyro: InvenSense Inc.

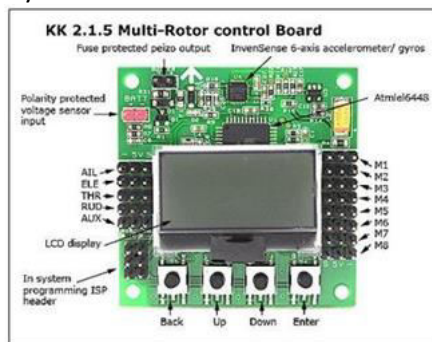


Figure:7 Multi-rotor control board

ELECTRONIC SPEED CONTROLLER (ESC)

An electronic speed controller (ESC) is an electronic circuit with the purpose to vary an electric motor's speed, its direction and possibly also to act as a dynamic brake. ESCs are often used on electrically powered radio controlled models, with the variety most often used for brushless motors essentially providing an electronically generated three-phase electric power low voltage source of energy for the motor.

We have used 60A electronic speed controllers to control each brushless motor in this experiment which can constantly supply required current to

drive brushless motors. It has the following specifications:

Constant Current: 60A

Burst Current: 80A

BRUSHLESS MOTOR

Brushless DC electric motor (BLDC motors, BL motors) also known as electronically commutated motors (ECMs, EC motors) are synchronous motors which are powered by a DC electric source via an integrated inverter/switching power supply, which produces an AC electric signal to drive the motor (AC, alternating current, does not imply a sinusoidal waveform but rather a bi-directional current with no restriction on waveform); additional sensors and electronics control the inverter output amplitude and waveform and frequency (i.e. rotor speed).

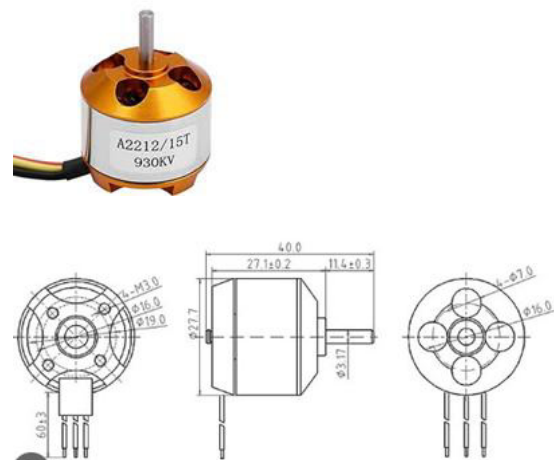


FIGURE: 8 Brushless Motor

LITHIUM POLYMER (LI-PO) BATTERY

LiPo batteries (short for Lithium Polymer) are a type of rechargeable battery that have become very popular because of their power to rate ratio. In other words, more electricity in a lighter package. Obviously, this is ideal for anything you're trying to get to fly. Another advantage is that LiPo's have a high discharge rate – which means they can deliver large amounts of power at once.



FIGURE: 9 Li-Po Battery



Specifications:

- BatteryConfiguration:11.1V2200mAh3cell
- BatteryCapacity:2200mAh
- MaxContinuousDischarge(C-rate/current):20CMaxBurst(3Sec)
(C-rate/current):45C

AnRFModule(RadioFrequencyModule)isusual
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Figure:10 Frequencytransmitter

Receiverspecification:

- Channel:6
- Frequencyband:2.4GHz
- Powerresource:1.5V*4"AA"battery
- Programtype:GFSK
- Weight:25G

Schematic of reaction torques on each motor of a quadcopter aircraft, due to spinning rotors.

Rotors1and3spininonedirection,whilerotors2and4spinintheoppositedirection,yieldingoppositngtorquesforcontrol.

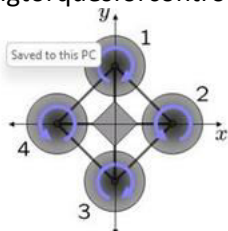
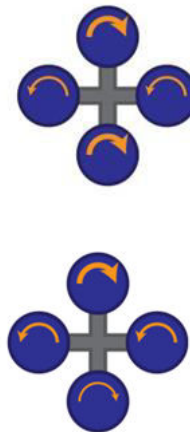
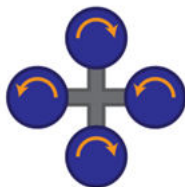


Figure:11Spinningdirection



Workingprincipleandworking

In this study, the aircraft was assumed to be a drone, which means it also stores the energy in the battery during the flying period. In this way, both flying the drone by using solar energy and storing the energy in the battery in order to extend the flight time can be achieved. The principle in which the drone works on it is NEWTONS

THIRD LAW OF MOTION.



Figure:12Pitchmotion

RESULTS

The testing done includes both the testing with different Payload, the testing of the Quadcopter. So in order for the testing to go on many things have to be done which includes

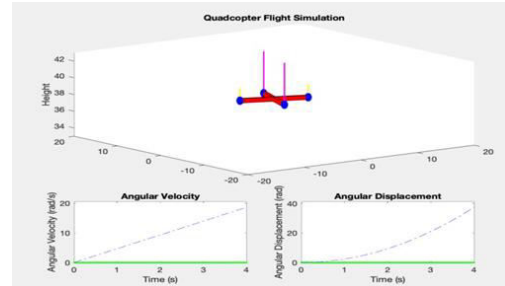


Figure:13testingwithdifferentPayload

ApplicationPerformanceManagement(APM)has to stabilize and when this is done you hear the beep sound telling you that the Quadcopter is ready. After that the APM is armed. The throttle is increased slightly. Then on the transmitter it should be in channel 1. After it is switched to channel 2 which



autonomously fly.

After that the return to launch mode helps it return to origin

CONCLUSION

In conclusion, this study has successfully explored the design and development of a human-carrying quadcopter, emphasizing its potential as a transformative solution for urban air mobility. Through comprehensive analysis and testing, the research has demonstrated that integrating advanced aerodynamic principles, lightweight materials, and innovative propulsion systems can result in a safe, efficient, and reliable aerial vehicle capable of transporting passengers. The findings reveal that while significant challenges, such as regulatory compliance, safety assurance, and technical limitations, must still be addressed, the advancements made in flight control algorithms and autonomous navigation systems are paving the way for practical implementation. Furthermore, this research underscores the importance of continued innovation and collaboration among engineers, regulators, and urban planners to facilitate the adoption of aerial transportation technologies. As urban populations continue to grow, the need for efficient transit solutions becomes increasingly urgent. The insights gained from this study not only contribute to the development of human-carrying quadcopters but also enhance the overall understanding of aerial mobility systems, potentially influencing the future landscape of transportation in urban environments. Future research should focus on further optimizing performance, enhancing safety measures, and exploring integration with existing transportation infrastructures to fully realize the capabilities of human-carrying quadcopters in modern urban settings.

REFERENCES

CFD study of an annular-ducted fan lift system for VTOL aircraft 2015 by Yun Jiang, Bo Zhang & Tao Huang https://www.researchgate.net/publication/282350303_CFD_study_of_an_annul

Quadcopter thrust optimization with ducted-

eISSN 1303-5150

propeller 2017 by Endorsees Kuantama & Radu Catalin Tarca https://www.researchgate.net/publication/320285199_Quadcopter_thrust_optimization_with_ducted-propeller

How ducting a propeller increases efficiency and

thrust <https://www.youtube.com/watch?v=Cew5JF8q6eY>

Standardization December 2018 roadmap for Unmanned Aircraft Systems, Version 1.0 Prepared by the ANSI Unmanned Aircraft Systems Standardization Collaborative (UASSC)

Brushless DC Motor vs. AC Motor vs. Brushed Motor in oriental

motor. <https://www.orientalmotor.com/brushless-dc-motors-gear-motors/technology/AC-brushless-brushed-motors.html>

Brushless Motors vs. Brushed Motors - What's the

Difference <https://www.thomasnet.com/articles/machinery-tools-supplies/brushless-motors-vs-brushed-motors/>

https://www.amazon.com/RSII2206-Brushless-Support-Compatible-Quadcopter/product?reviews/B07HJ3C5TR?reviewerType=all_reviews

All About a Multirotor FPV Drone Power Distribution Board | Getup

Learn <https://www.getfpv.com/learn/new-to-fpv/all-about-multirotor-fpv-drone-power-distribution-board/>
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